## IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

In re Application of:

Examiner: E. Miller

Karabeyoglu, et al.

Group Art Unit: 3641

Serial No.: 09/505,516

Filed: February 17, 2000

San Francisco, CA 94111

For:

HIGH REGRESSION RATE

HYBRID ROCKET PROPELLANTS AND

METHOD OF SELECTING

Date: July 24, 2001

<u>CERTIFICATE OF FACSIMILE TRANSMISSION</u>

I hereby certify that this correspondence is being faxed to Facsimile No. 703-305-7687 to the attention of Examiner Edward A. Miller, Group Art Unit 3641, at the Office of the Assistant Commissioner for Patents, Washington, D.C. 20231 on July 24, 2001.

Claudia Galik

## **AMENDMENT**

**Assistant Commissioner for Patents** Washington, D.C. 20231

Sir:

This is in response to the Office Action dated March 8, 2001 in the abovereferenced application. Please amend the application as follows:

## IN THE CLAIMS:

Please amend the claims as follows:

14. (Twice Amended) A method of combusting a propellant that exhibits desirable regression rate during combustion within a port having a gas stream flowing through the port, comprising the steps of:

providing a propellant having under heat transfer from the gas stream flowing though the port, a liquid layer with surface tension o and liquid viscosity  $\mu_1$  values that promote entrainment of droplets from said liquid layer into said gas stream flowing in said port, and said

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